

CR 54393



(NASA-CR-54393) GENERATION OF LONG TIME
CREEP DATA ON REFRactory ALLOYS AT
ELEVATED TEMPERATURES Quarterly Report,
26 Dec. 1964 - 26 Mar. 1965 (TRW
Equipment Labs.) 41 p

N73-72201

Unclassified
00/99 68574

GENERATION OF LONG TIME CREEP DATA ON REFRACTORY ALLOYS AT ELEVATED TEMPERATURES

SEVENTH QUARTERLY REPORT



Prepared for

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION
LEWIS RESEARCH CENTER
UNDER CONTRACT NAS 3-2543

TRW EQUIPMENT LABORATORIES
CLEVELAND, OHIO

JUN 1 1965

Seventh Quarterly Report

for

December 26, 1964 to March 26, 1965

GENERATION OF LONG TIME CREEP DATA
OF REFRACTORY ALLOYS AT ELEVATED TEMPERATURES

Prepared by:

J. C. Sawyer and C. H. Philleo

Approved by:

E. A. Steigerwald

Prepared for:

National Aeronautics and Space Administration
Contract No. NAS 3-2545

Technical Management

Paul E. Moorhead
NASA - Lewis Research Center
Space and Power Systems

April 28, 1965

Materials Research and Development Department
TRW EQUIPMENT LABORATORIES
TRW Inc.
23555 Euclid Avenue
Cleveland, Ohio 44117

NOTICE

This report was prepared as an account of Government-sponsored work. Neither the United States, nor the National Aeronautics and Space Administration (NASA), nor any person acting on behalf of NASA:

- a) Makes any warranty or representation, expressed or implied, with respect to the accuracy, completeness, or usefulness of the information contained in this report, or that the use of any information, apparatus, method, or process disclosed in this report may not infringe privately-owned rights; or
- b) Assumes any liabilities with respect to the use of, or for damages resulting from the use of any information, apparatus, method, or process disclosed in this report.

As used above, "person acting on behalf of NASA" includes any employee or contractor of NASA, or employee of such contractor, to the extent that such employees or contractor of NASA or employee of such contractor prepares, disseminates, or provides access to, any information pursuant to his employment of contract with NASA, or his employment with such contractor.

Requests for copies of this report should be referred to:

National Aeronautics and Space Administration
Office of Scientific and Technical Information
Washington 25, D. C.

Attention: AFSS-A

FOREWORD

The work described herein is being performed by TRW Inc. under the sponsorship of the National Aeronautics and Space Administration under Contract NAS 3-2545. The purpose of this study is to obtain design creep data on refractory metal alloys for use in space power systems.

The program is administered for TRW Inc. by E. A. Steigerwald, Program Manager. J. Sawyer is the Principle Investigator. H. Philleo and R. Ebert contributed to the program.

ABSTRACT

Ultra-high vacuum creep data are presented for tungsten, tungsten-25% rhenium, TZM and TZC molybdenum alloys, and AS-30 columbium alloy. TZC alloy exhibits the greatest creep strength of the turbine alloys examined and has been selected for the initial 10,000 hour creep tests.

TABLE OF CONTENTS

	<u>Page No.</u>
INTRODUCTION.....	1
INTERNATIONAL SYSTEM OF UNITS.....	1
MATERIALS & PROCEDURE.....	2
EXPERIMENTAL RESULTS.....	2
TUNGSTEN AND TUNGSTEN-25% RHENIUM.....	2
MOLYBDENUM BASE ALLOYS - TZM AND TZC.....	8
COLUMBIUM ALLOY - AS-30.....	14
FUTURE WORK.....	14
APPENDIX I.....	16
APPENDIX II.....	18

INTRODUCTION

The object of this program is to obtain long-time creep data on selected refractory alloys which have potential use in advanced space power systems. Earlier reports have described the equipment design and test methods, and have presented creep data on tungsten, tungsten-25% rhenium alloy, TZM molybdenum alloy, and AS-30 columbium alloy. This report presents additional creep data for the above materials and for TZC molybdenum alloy.

International System of Units

Certain of the field centers of the National Aeronautics and Space Administration have followed the National Bureau of Standards in adopting the units of the International System (SI). As a result, in this report the International System of units will be presented along with the more conventional engineering units.

The conversion values for the terms used most frequently in this program are presented in the following table.

TABLE I
COMMONLY USED CONVERSIONS FROM ENGLISH
TO INTERNATIONAL UNITS

<u>Temperature</u>	<u>English</u>	<u>International</u>
1800°F	=	982° Celsuis (°C)
2000°F	=	1093°C
2200°F	=	1204°C
2850°F	=	1566°C
3092°F	=	1700°C
3200°F	=	1760°C
Stress - 1000 psi	=	6.895×10^6 newtons/meter ²
Vacuum - 1×10^{-8} Torr (mmHg)	=	1.333×10^{-6} newtons/meter ²

Materials and Procedure

A summary of the material variables being evaluated is presented in Table 2, while the compositions of the specific alloys are summarized in Table 3. A detailed description of the processing history of tungsten, tungsten-25% rhenium, TZM, TZC, and AS-30 was presented in the previous progress report(1)*, and similar data for Sylvania A alloy are given in Appendix 1 of this report.

The geometries for both the rod and sheet specimens are shown in Figures 1 and 2. The test procedure involved obtaining a vacuum of 5×10^{-10} Torr ($6.65 \times 10^{-8} \text{ N/m}^2$) or better at room temperature, then heating the specimen at a rate so that the pressure never rises above 1×10^{-5} Torr ($1.33 \times 10^{-4} \text{ N/m}^2$). The specimen is held at temperature for approximately two hours prior to load application. After the first few minutes of load application, specimen contraction can often be observed due to a slight temperature decrease produced by the increased grip contact and the resulting increase in heat conduction. As the test proceeds the vacuum continuously improves and approaches the 10^{-9} to 10^{-10} Torr range (10^{-7} to 10^{-8} N/m^2 range).

Experimental Results

The general test plan involves conducting 1000 hour screening tests on all the alloys and from these data selecting appropriate materials for long time creep tests. At present, the test end point has been chosen as a total creep of 1% for the materials applicable to turbines (AS-30, Cb132M, TZM, TZC) and 5% creep for the cladding alloys (W, W-25%Re). The basic creep data for all the tests completed or initiated during this quarter are presented in Appendix 2.

Tungsten and Tungsten-25% Rhenium

The creep curves for tungsten and tungsten-25% rhenium tested at 3200°F (1760°C) are presented in Figure 3. The W-25% Re alloy loaded at 1500 psi ($1.03 \times 10^7 \text{ N/m}^2$) showed a total extension of 2.76% in 800 hours while the tungsten loaded at 1000 psi ($6.9 \times 10^6 \text{ N/m}^2$) extended 1.02% in 688 hours.

* Numbers in parentheses pertain to references in the Bibliography.

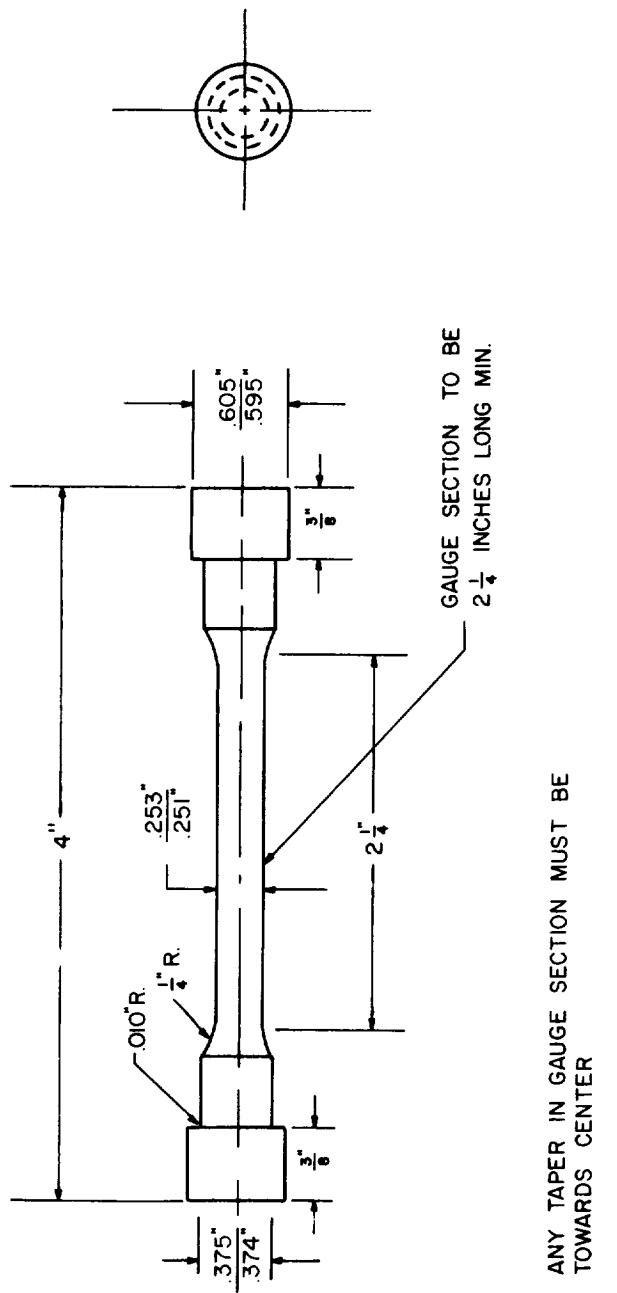
TABLE 2
SUMMARY OF MATERIAL VARIABLES BEING EVALUATED IN CREEP PROGRAM

Material	Form	Test Temperature	Test Condition
Tungsten	Arc-Melted 0.030" Sheet	3200°F (1760°C)	Recrystallized 2 hours, 3200°F (1760°C)
Tungsten-25% Rhenium	Arc-Melted 0.030" Sheet	3200°F (1760°C)	Recrystallized 2 hours, 3200°F (1760°C)
Sylvania A	Powder Metallurgy 0.030" Sheet	3200°F (1760°C)	Recrystallized 2 hours, 3200°F (1760°C)
AS-30	3/4" Plate	2000-2200°F (1093-1204°C)	As-received, stress-relieved condition, (R _c 29)
Cb132M	3/4" Plate	2000°F (1093°C)	Recrystallized 1 hour, 3092°F (1700°C)
TZM	"Pancake" Forging	2000°F (1093°C)	(Cond. 1) As-received, stress-relieved condition (R _c 32) (Cond. 2) Recrystallized 1 hour, 2850°F (1566°C)
TZC	3/4" Plate	1800, 2000, 2200°F (982, 1093, 1204°C)	Recrystallized 1 hour, 3092°F (1700°C)
T-2222*	0.030" Sheet	1800, 2000, 2200°F (982, 1093, 1204°C)	As-received

* Originally scheduled to be tested as ST-222 plate material, program plan revised to include material as T-222 grade applicable for tubing.

TABLE 3
CHEMICAL COMPOSITION OF ALLOYS BEING EVALUATED IN CREEP PROGRAM (WEIGHT %)

Material	W	Re	Cb	Mo	Ta	Hf	C	N ₂	T ₁	Zr	Ni	O ₂	H ₂
Tungsten	Bal												
Tungsten-25% Rhenium	Bal	24.9											
Sylvania A	Bal						0.52	.030					
AS-30	Bal												
Cb-132M	15.0	Bal.	5.07	19.8									
TZM		Bal											
TZC		Bal											
ST-222 (T-222)	10.4	Bal	2.47				.0086						



NOTE: ANY TAPER IN GAUGE SECTION MUST BE
TOWARDS CENTER

ALL TOLERANCES $\pm .010$ " UNLESS OTHERWISE
NOTED

FIG. I: CREEP SPECIMEN USED FOR DISC AND PLATE STOCK.

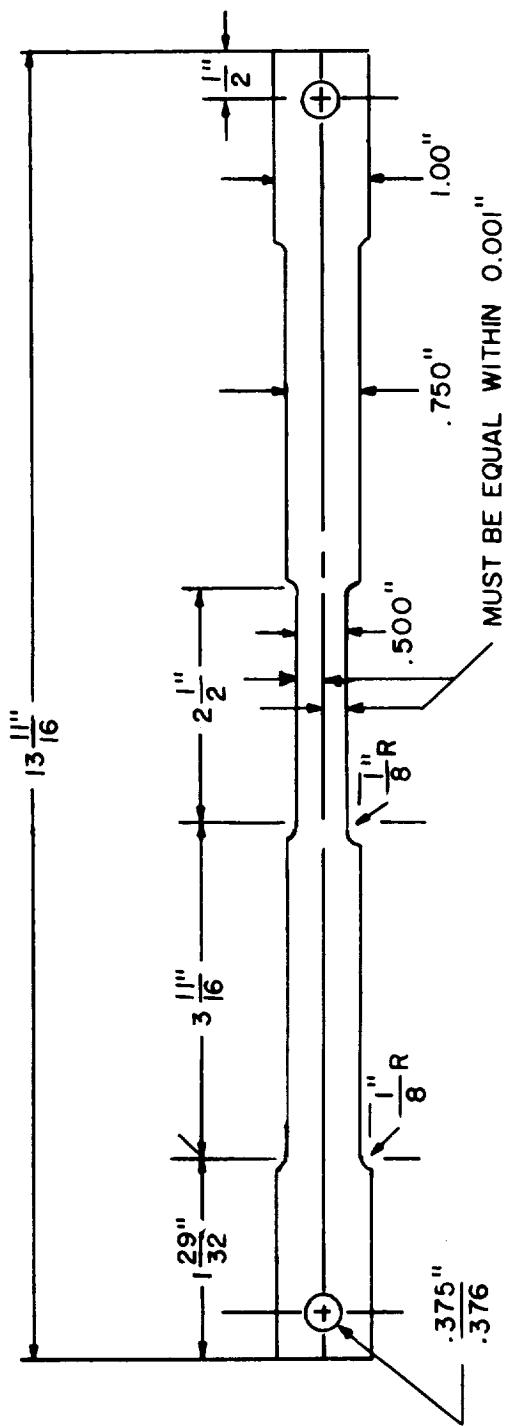


FIG. 2: CREEP SPECIMEN USED FOR SHEET STOCK

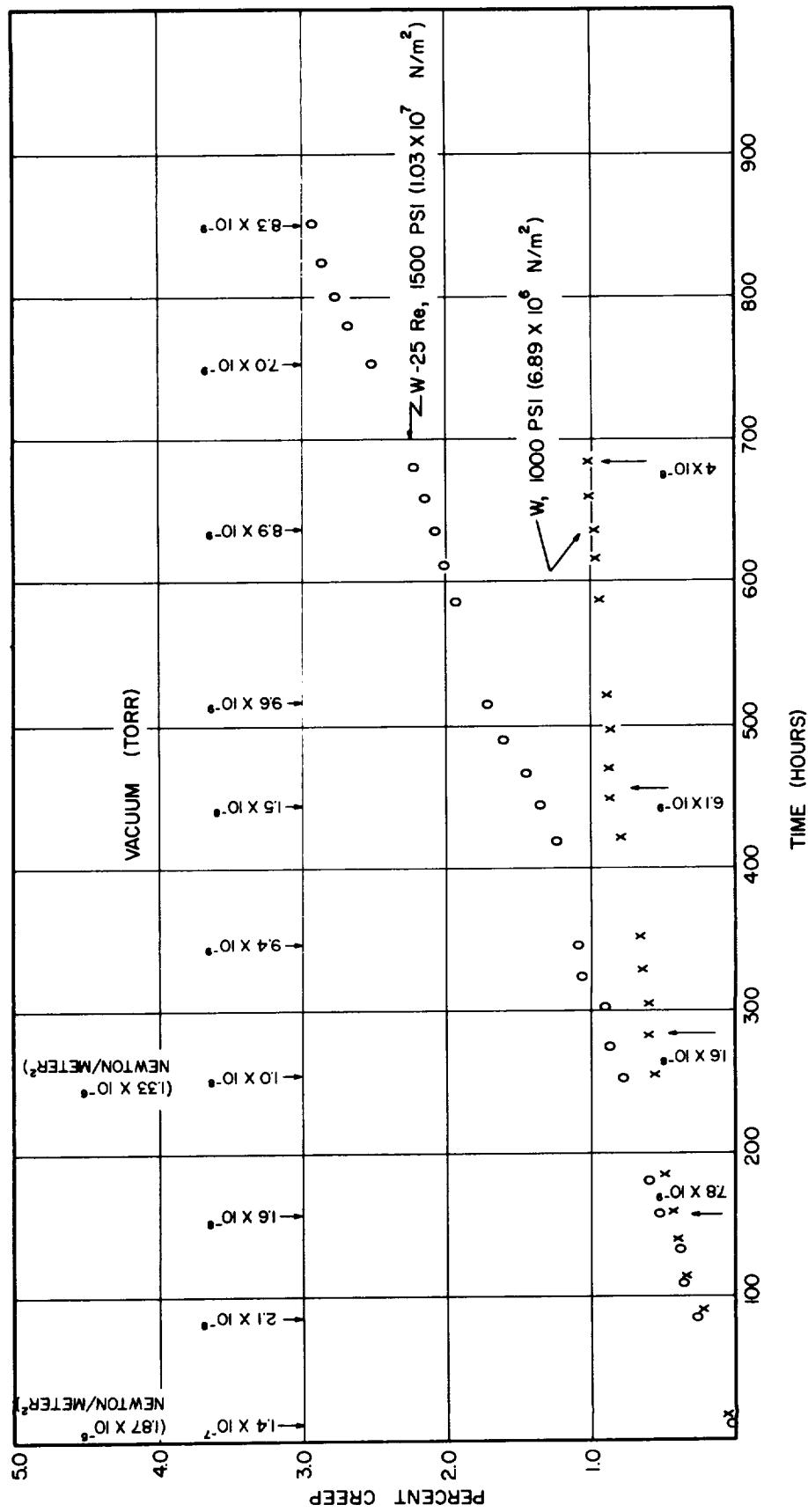


FIG. 3: CREEP OF TUNGSTEN AND TUNGSTEN-25% RHENIUM ALLOYS TESTED AT 3200°F,
(1760°C).

For purposes of initially comparing the available data, the Larson-Miller plot was employed with a value of 15 for the constant (see Figure 4). Using this method of presentation, no significant difference between the creep properties of the tungsten and tungsten-25% rhenium was apparent at the lower levels of applied stress.

In general, when published data obtained at higher temperatures on powder metallurgy tungsten in an argon atmosphere were analyzed, they indicated a higher stress capability for a given value of the Larson-Miller parameter.

Molybdenum Base Alloys - TZM and TZC

The test data obtained on TZM at 2000°F (1093°C) under an applied stress of 10,000 psi ($6.89 \times 10^7 \text{ N/m}^2$) are presented in Figures 5 and 6. Two material conditions are being evaluated. The variable representing the as-received, stress-relieved condition (Figure 5) indicates a relatively constant creep rate over the 2000 hour test period, while the specimens from the recrystallized material exhibited creep behavior that was considerably different. At approximately 325 hours a significant discontinuity in the creep curve was apparent in the recrystallized material followed by an extended period where no creep and even some specimen contraction was measured. In many of the creep curves, obvious discontinuities have been noted and considerable effort has been devoted to determining whether these variations can be attributed to experimental problems or to real material behavior. A check of the temperature measurements using both the thermocouples contained in the system and the optical pyrometer have indicated that temperature fluctuations are not the cause. Variations in extensometer readings are also not believed to be the primary contributing factor, since all measurements are performed on the same instrument and the observed fluctuations are greater than differences noted between various observers. On this basis, the discontinuities are believed to be a material effect associated with some type of phase change which is occurring during testing. Analysis of specimens after testing will be performed in an effort to define the variations in material which may be occurring during the test exposure.

Since the TZM is being tested as pancake forgings, a question exists as to whether significant variations are present in the degree of working throughout the forging and their possible influence on creep properties. The variation in microstructure as a function of position from the outer edge of the forging is shown in Figure 7. The structures are reasonably consistent and comparable to those found in the specimen gage sections. The structure of the cross-section of the TZM is shown in Figure 8, which illustrates the typical flow pattern. The results indicate that variations in grain pattern can exist between the gage section and the specimen button-head; however, over the 2" gage length the structure is reasonably reproducible and should not introduce a wide degree of scatter in test results.

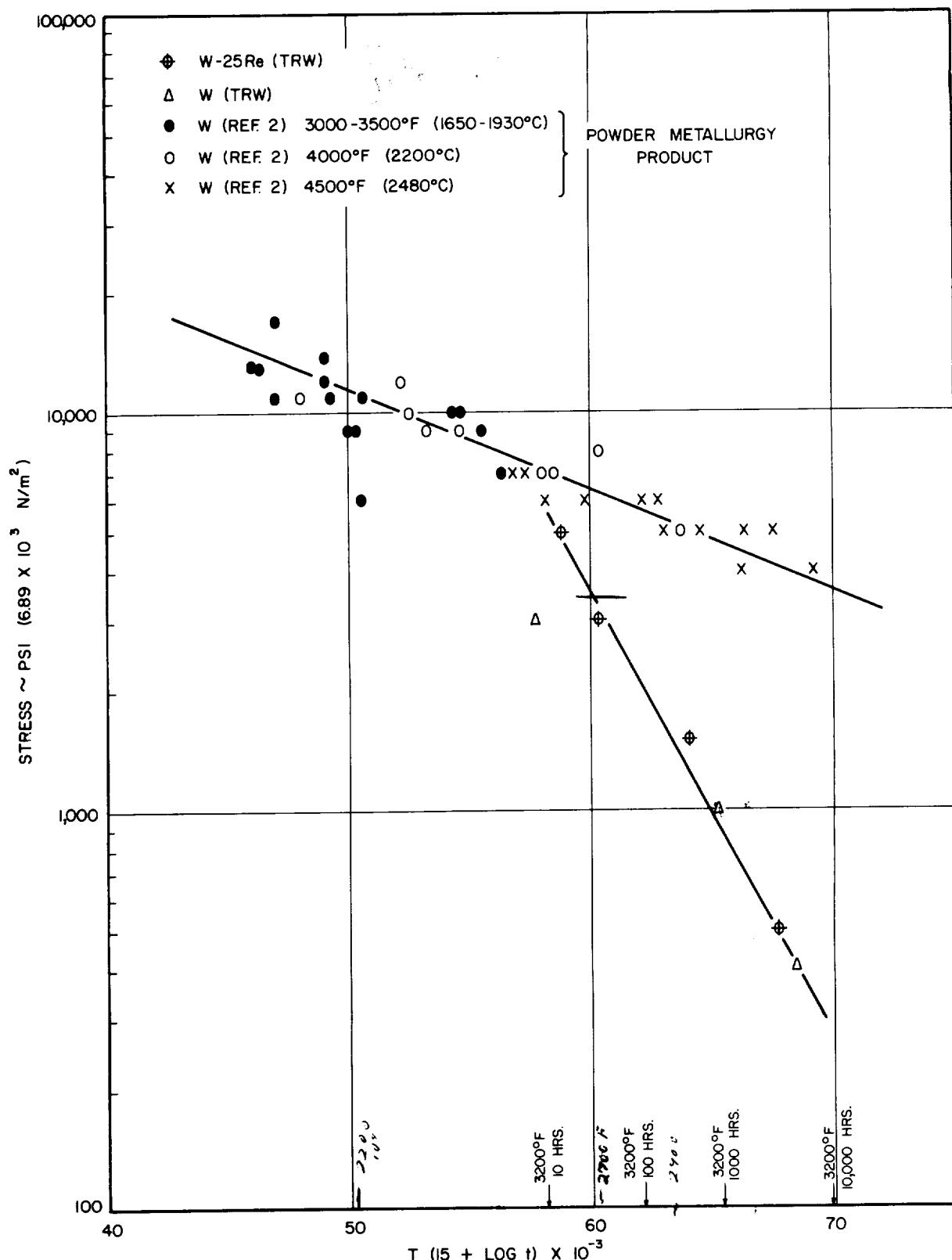


FIG. 4: LARSON-MILLER PLOT OF TUNGSTEN AND TUNGSTEN-25% RHENIUM 1% CREEP DATA, (T = TEST TEMPERATURE °R, t = TEST TIME, HOURS).

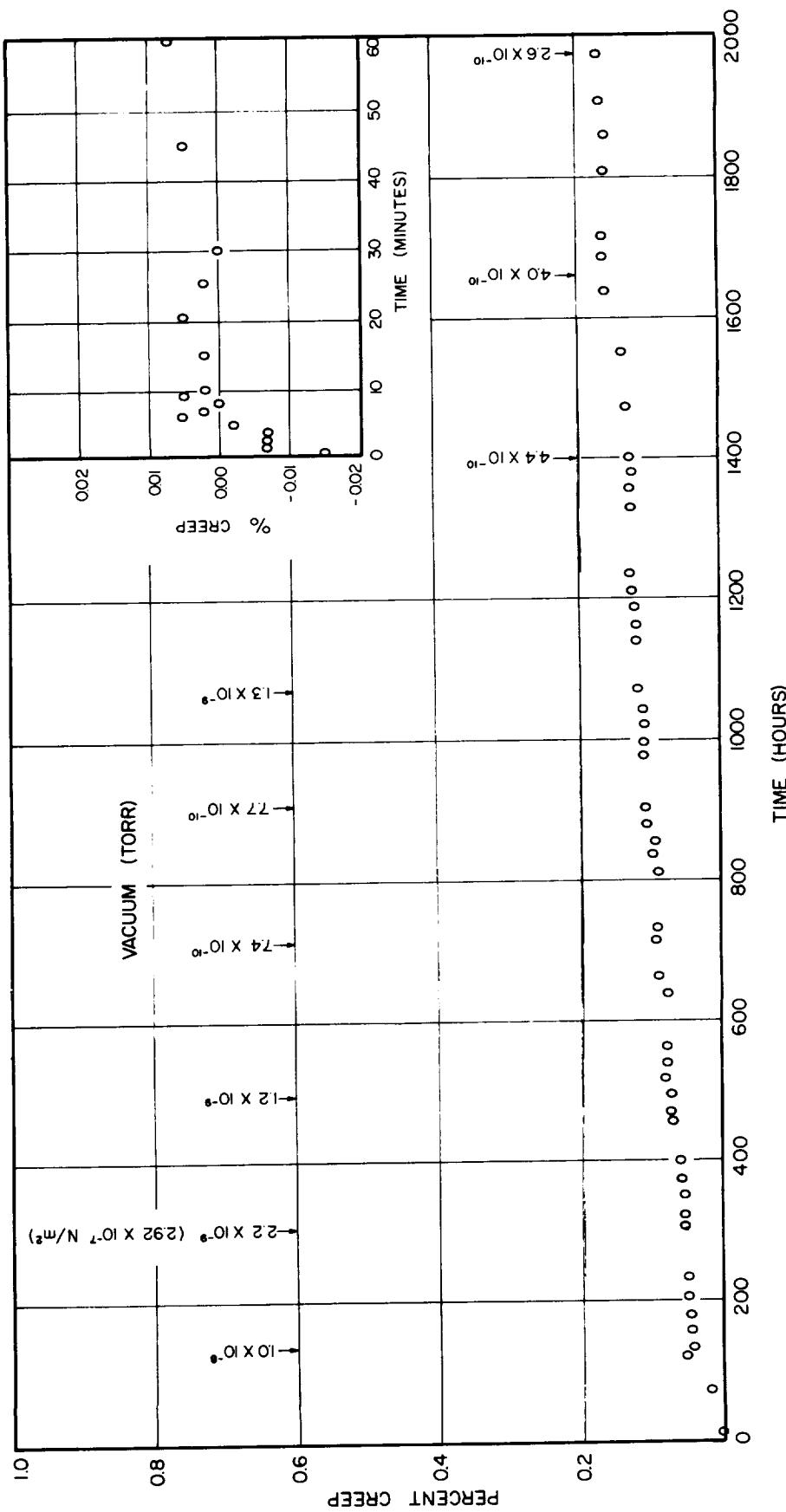


FIG. 5: CREEP OF STRESS-RELIEVED TZM TESTED AT 2000°F (1093°C), 10,000 PSI
 $(6.89 \times 10^7 \text{ N/m}^2)$.

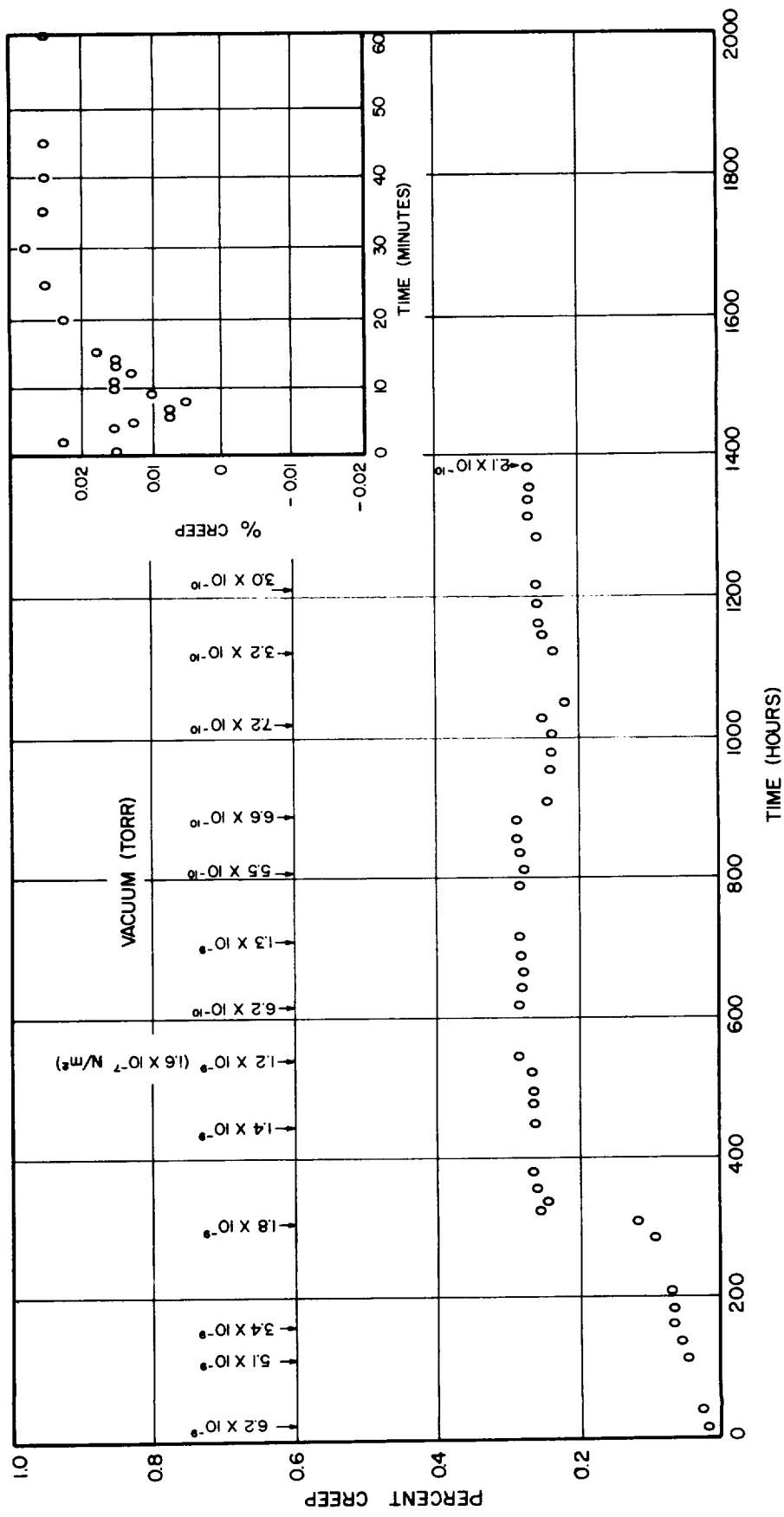
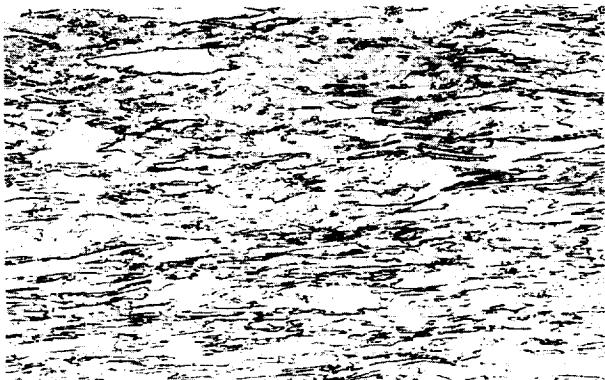


FIG. 6: CREEP OF TZM RECRYSTALLIZED AT 2850°F (1560°C), TESTED AT 2000°F (1093°C),
10,000 PSI ($6.89 \times 10^7 \text{ N/m}^2$).



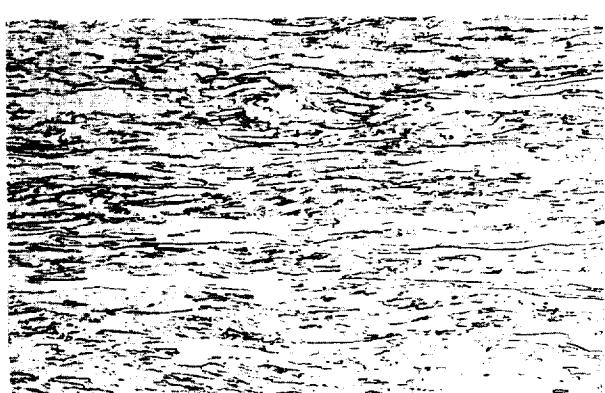
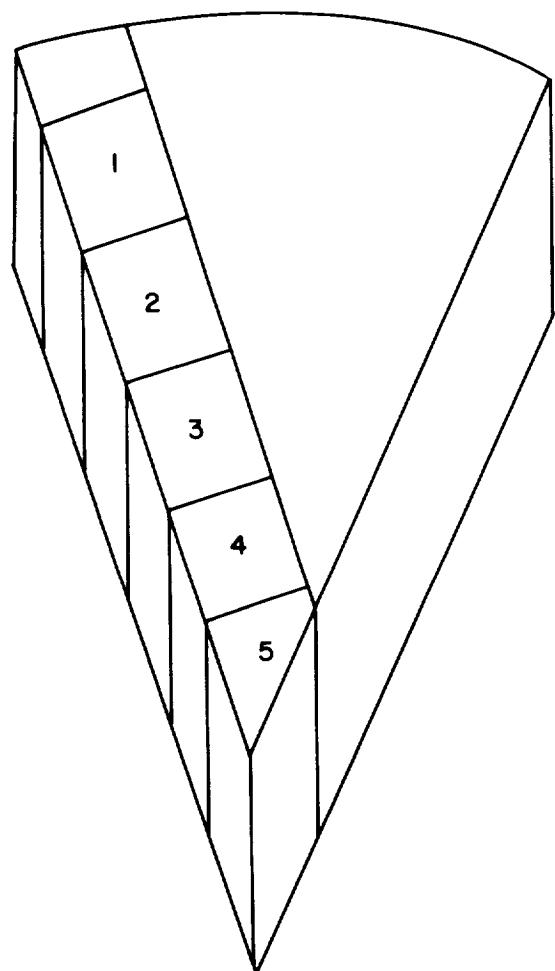
1

AN 866



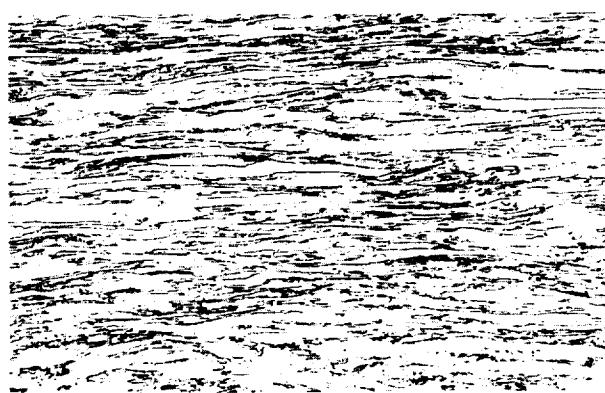
2

AN 865



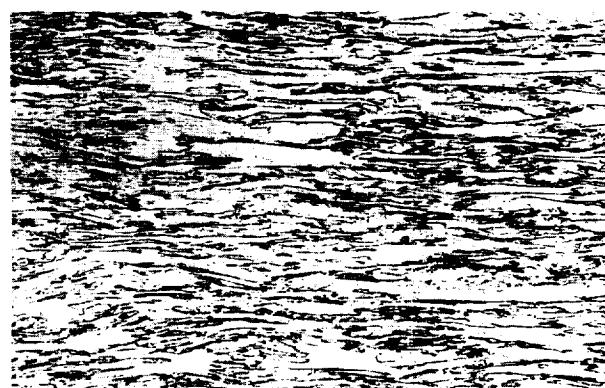
3

AN 864



4

AN 863



5

AN 862

**FIG. 7: PHOTOMICROGRAPHS SHOWING
ETCHED SECTIONS OF TZM PANCAKE
FORGING. 100X, MURAKAMI'S ETCH.
AS RECEIVED.**

CENTER OF FORGING



08305

OUTER EDGE OF FORGING

FIG. 8: PHOTOGRAPH OF MACROETCHED SECTION OF TZM PANCAKE - FORGED DISC. SHOWING FLOW LINES DUE TO FORGING. 1 1/2 X.

The initial test on TZC was performed at 2000°F (1093°C) at a load of 14,000 psi (9.65×10^7 N/m²). No measurable amount of creep was observed after 520 hours. In order to save set-up time and obtain an approximation of a suitable stress for the TZC alloy at this temperature, a "staircase" loading technique was employed. Thus the specimen was reloaded at 20,000 psi (1.38×10^8 N/m²) for 259 hours, and 22,000 psi (1.52×10^8 N/m²) for 572 hours. Between each of these tests the load was removed, the specimen was heated to 3092°F (1700°C) for 1 hour, and cooled to room temperature prior to retesting. A summary of the results obtained on the sequentially loaded specimen is shown in Figure 9. The repeated loading and heating of the specimen appeared to have improved the creep properties relative to the initial recrystallized TZC material.

On the basis of these preliminary data, stresses were selected for long-time tests on TZC at 2200°F (1204°C) and 2000°F (1093°C) and the current results are presented in Figure 10.

A comparison of the TZM and TZC data on the basis of the Larson-Miller parameter is given in Figure 11, along with previously published data obtained under less stringent vacuum conditions(3). The creep resistance of the TZM alloy used in the current program is substantially less than that previously reported. However, at present this difference cannot be associated with either material processing, composition, or test conditions.

Columbium Alloy - AS-30

The creep data obtained during this quarter on AS-30 are summarized in Figure 12. At the specific test temperatures, the material has less creep resistance than the molybdenum-base alloys. A presentation of the data on a Larson-Miller plot is shown in Figure 13 for various degrees of creep extension.

Future Work

During the next report period, additional 10,000 hour tests will be initiated with the TZC alloy. These tests will be at 1800°F (982°C) and 2200°F (1204°C).

The ST-222 tantalum alloy plate has been received; and, in agreement with NASA, this alloy was returned to the vendor to be rolled into T-222 sheet material since the actual composition of the as-received ST-222 was within the T-222 alloy specification. The reworked alloy should be received during the next report period, and long-time tests will be initiated.

The Sylvania "A" sheet received contained a number of defects; however, it was possible to obtain suitable test specimens. This material will be tested on a limited basis during the next report period, along with the Cb132M columbium alloy which is scheduled for delivery early in April.

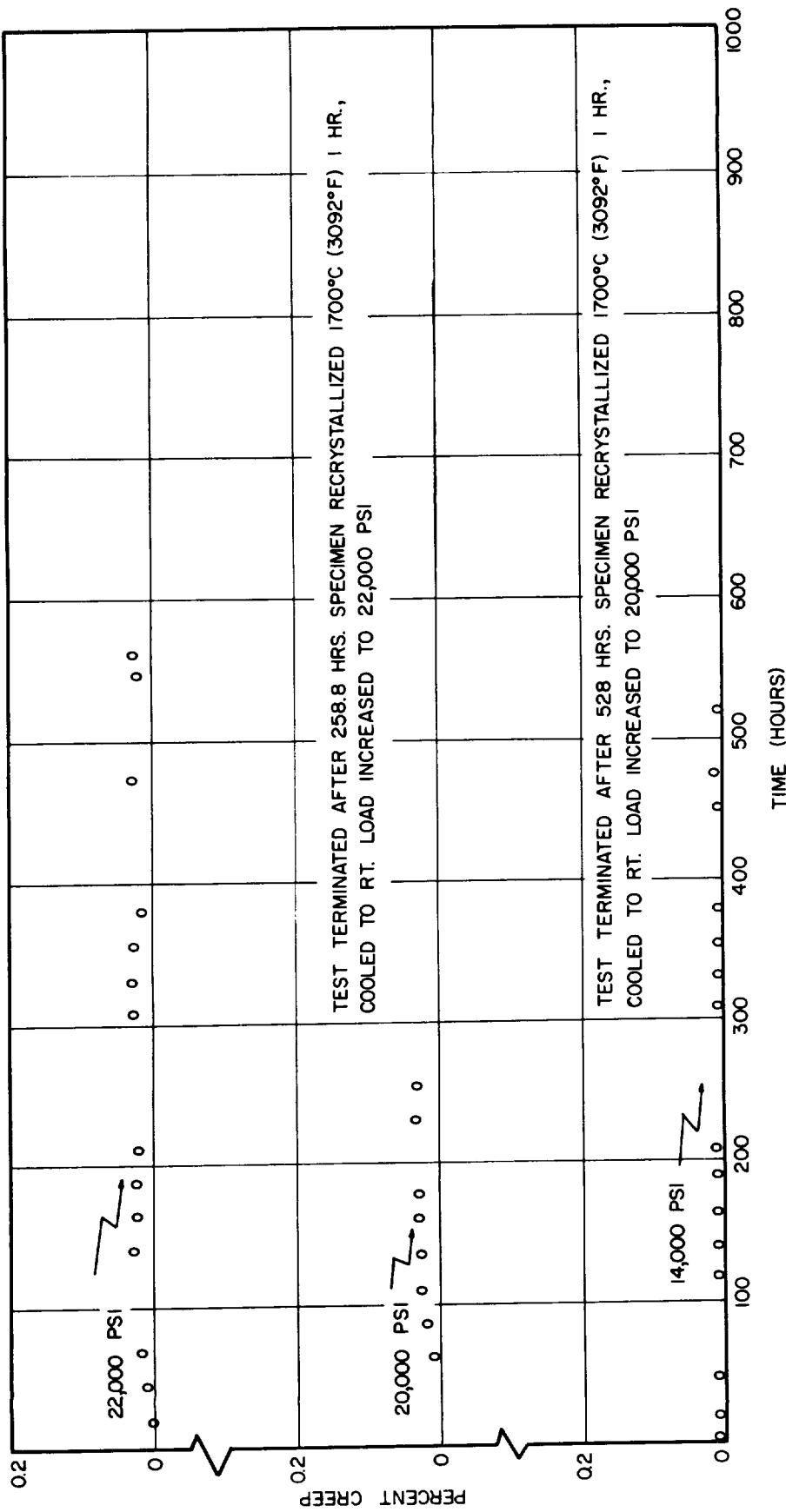


FIG. 9: CREEP OF RECRYSTALLIZED TZC SPECIMEN TESTED AT 2000°F (1093°C) AND STRESSES
OF 14,000 PSI (9.65×10^7 N/m 2), 20,000 PSI (1.38×10^8 N/m 2),
22,000 PSI (1.52×10^8 N/m 2) VACUUM ENVIRONMENT (< 10^{-8} TORR).

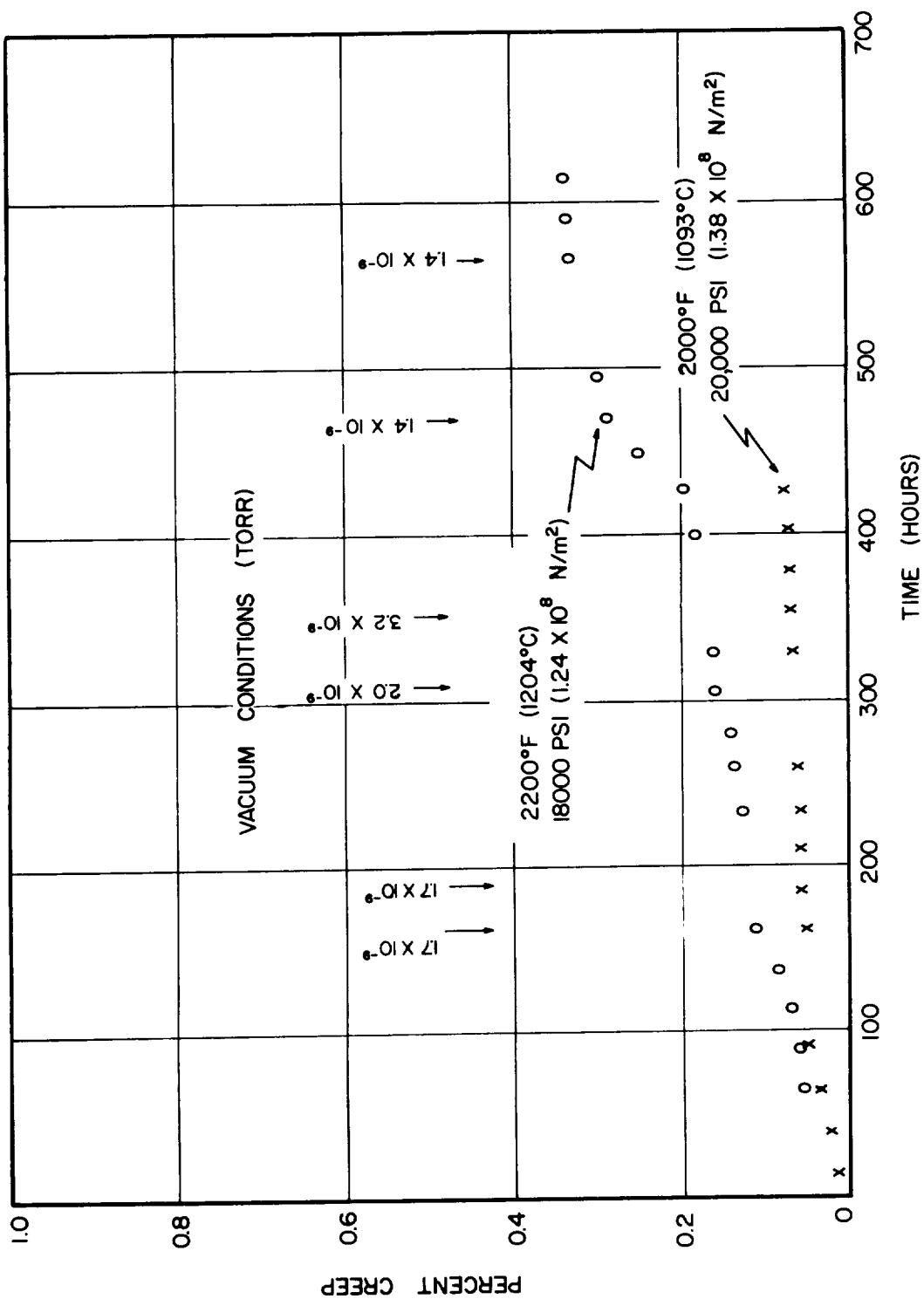


FIG. 10: CREEP OF RECRYSTALLIZED TZC ALLOY, VACUUM ENVIRONMENT
 $< 10^{-8}$ TORR (1.33×10^{-8} N/m²).

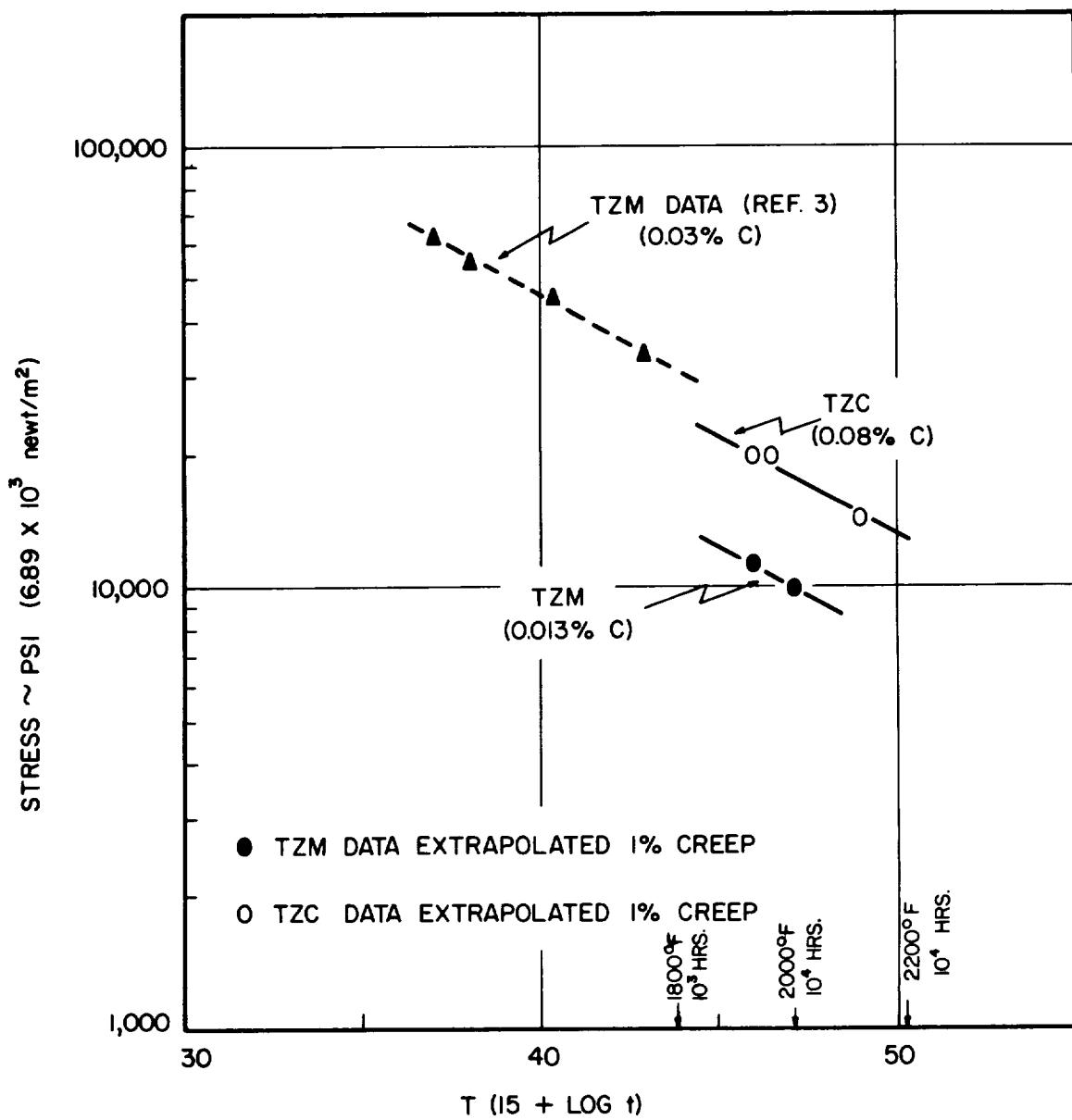


FIG. II: COMPARISON OF 1% CREEP DATA FOR MOLYBDENUM ALLOYS OBTAINED UNDER ULTRA HIGH VACUUM CONDITIONS WITH PREVIOUSLY PUBLISHED DATA.

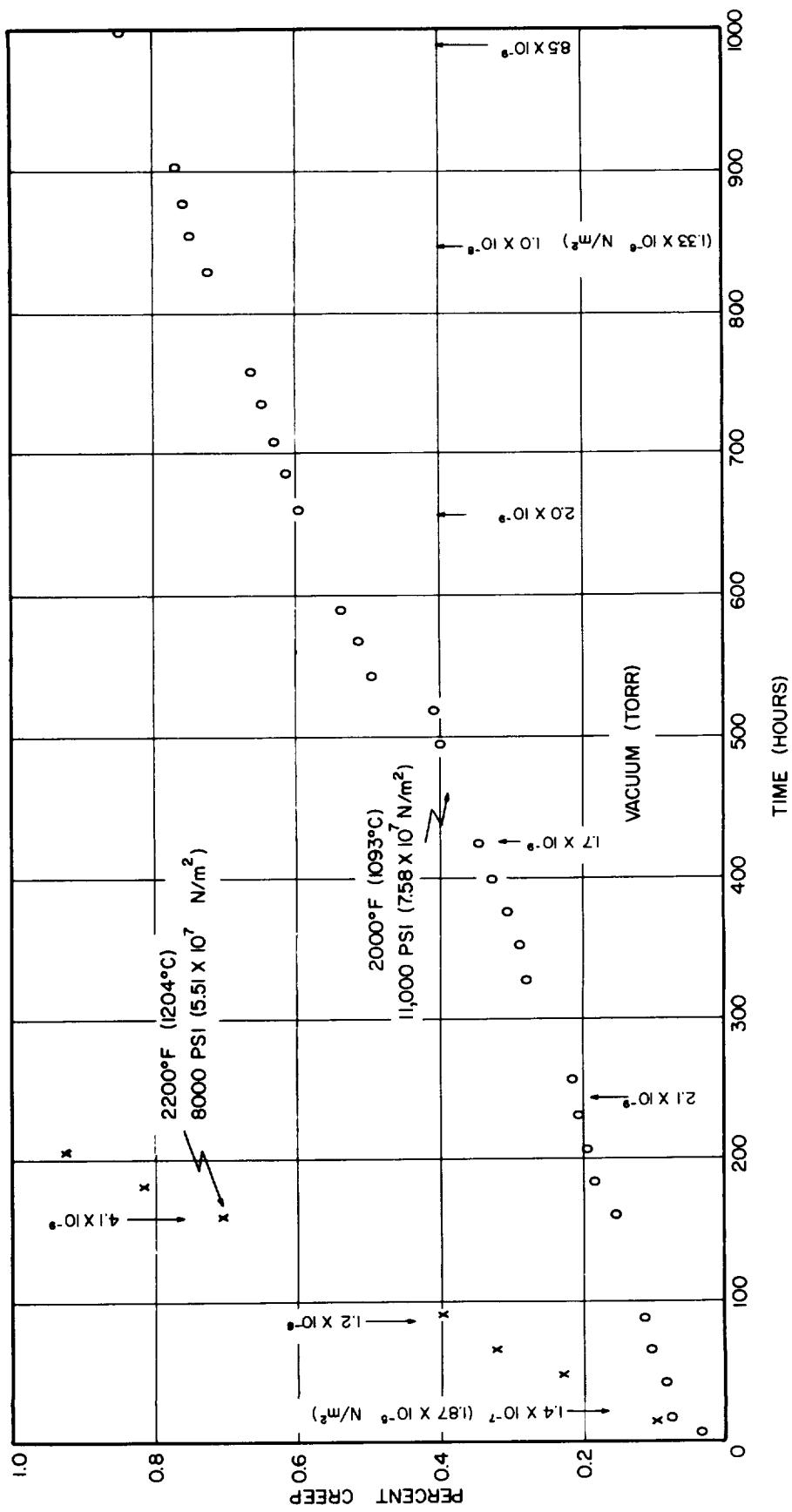


FIG. I2: CREEP OF STRESS-RELIEVED AS-30 PLATE.

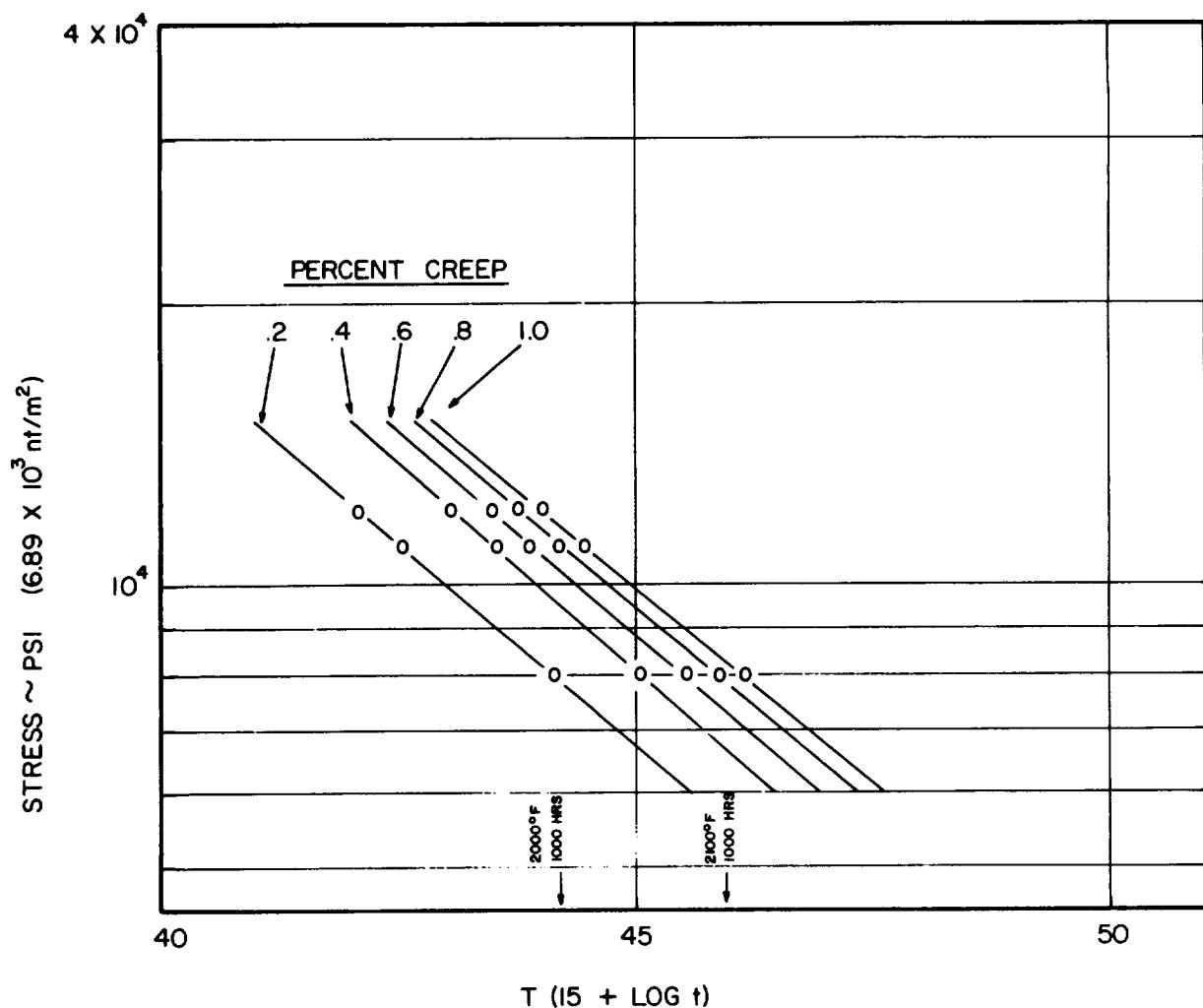


FIG. 13: LARSON MILLER PLOT OF CREEP DATA FOR AS-30
STRESS-RELIEVED PLATE TESTED UNDER ULTRA-
HIGH VACUUM.

BIBLIOGRAPHY

1. J. C. Sawyer, and C. H. Philleo, "Generation of Long Time Creep Data on Refractory Alloys at Elevated Temperatures", Sixth Quarterly Report CR 54287, NASA Contract NAS 3-2545, (January 15, 1965).
2. F. F. Schmidt and H. R. Ogden, "The Engineering Properties of Tungsten Alloys", DMIC Report 191, (September 27, 1963).
3. Creep Rupture Properties of Stress Relieved TZM Alloy - SNAP 50/SPUR Program Part 1. Report No. PAL TDR-64-116, October 1964.

TABLE IPROCESSING OF SYLVANIA "A" SHEET

Vendor: Sylvania Electric Products Inc.
Chemical and Metallurgical Division
Towanda, Pennsylvania

Processing History:

1. Rolling slabs were made by isostatically pressing powder
2. Slabs rolled at 1500-1900°C to 0.032". Total reduction 90%
3. Intermediate annealing - None
4. Final stress relief - five minutes at 150°C
5. Sheet trimmed with an abrasive saw and chemically cleaned.

Hardness: 738 DPH
62 R_c

APPENDIX II

CREEP DATA

TABLE ICREEP TEST DATA, W-25%Re SHEET, TESTED AT 3200°F (1760°C), 1500 PSI ($1.03 \times 10^7 \text{ N/m}^2$)

Time (Minutes)	Length Change			Time (Hours)	Length Change		
	ΔL (in) (2" G.L.)	Creep (%)	Pressure (Torr)		ΔL (in) (2" G.L.)	Creep (%)	Pressure (Torr)
5	.00005	.0025	3.5×10^{-7}	10.6	.00005	.0025	1.4×10^{-7}
10	.00010	.005		82.7	.00450	.225	2.1×10^{-8}
15	.00010	.005		106.6	.00700	.350	2.0×10^{-8}
20	.00010	.005		130.6	.00735	.3675	1.6×10^{-8}
30	.00010	.005		154.6	.01065	.5325	1.6×10^{-8}
45	.00010	.005		178.8	.01170	.585	1.4×10^{-8}
60	.00010	.005		250.6	.01535	.767	1.0×10^{-8}
				274.5	.01730	.865	9.8×10^{-9}
				298.6	.01795	.897	1.0×10^{-8}
				322.5	.02135	1.067	9.6×10^{-9}
				346.7	.02195	1.097	9.4×10^{-9}
				418.5	.02480	1.24	8.5×10^{-9}
				442.9	.02700	1.35	1.5×10^{-8}
				466.7	.02910	1.455	8.4×10^{-9}
				490.7	.03225	1.612	1.0×10^{-8}
Specimen S-8	515.0	.03345		515.0	.03345	1.672	9.6×10^{-9}
	586.7	.03870		586.7	.03870	1.935	1.0×10^{-8}
	611.1	.04010		611.1	.04010	2.005	9.2×10^{-9}
	634.6	.04140		634.6	.04140	2.07	8.9×10^{-9}
	658.7	.04300		658.7	.04300	2.15	1.2×10^{-8}
	682.7	.04490		682.7	.04490	2.245	7.9×10^{-9}
	754.7	.05065		754.7	.05065	2.532	7.0×10^{-9}
	781.3	.05385		781.3	.05385	2.692	7.5×10^{-9}
	802.6	.05525		802.6	.05525	2.762	7.2×10^{-9}
	826.6	.05700		826.6	.05700	2.85	7.2×10^{-9}
	850.5	.05835		850.5	.05835	2.917	8.3×10^{-9}

Test in progress

TABLE IICREEP TEST DATA, TUNGSTEN SHEET, TESTED AT 3200°F (1760°C), 1000 PSI (6.89×10^6 N./m²)

<u>Time (Minutes)</u>	<u>Length Change ΔL (in) (2" G.L.)</u>	<u>Creep (%)</u>	<u>Pressure (Torr)</u>	<u>Time (Hours)</u>	<u>Length Change ΔL (in) (2" G.L.)</u>	<u>Creep (%)</u>	<u>Pressure (Torr)</u>
1	.00005	.0025	4.4×10^{-8}	16.6	.0018	.090	2.2×10^{-8}
2	.00015	.0075	88.4	.0048	.240	1.0×10^{-8}	
3	.00015	.0075		.00615	.307	1.1×10^{-8}	
4	.00010	.005	136.3	.00750	.375	9.2×10^{-9}	
5	.00015	.0075	160.3	.00880	.440	7.8×10^{-9}	
6	.00020	.010	184.4	.00960	.480	6.9×10^{-9}	
7	.00020	.010	256.3	.01115	.557	5.6×10^{-8}	
8	.00020	.010	280.7	.01190	.595	1.6×10^{-8}	
9	.00020	.010	304.5	.01215	.607	3.9×10^{-9}	
10	.00015	.0075	328.4	.01265	.632	6.8×10^{-9}	
15	.00020	.010	352.7	.01330	.665	6.5×10^{-9}	
20	.00010	.005	424.5	.01575	.787	7.2×10^{-9}	
25	.00020	.010	448.9	.01725	.862	6.1×10^{-9}	
30	.00020	.010	472.3	.01705	.852	6.3×10^{-9}	
60	.00020	.010	496.4	.01725	.862	9.0×10^{-9}	
			520.4	.01740	.870	4.6×10^{-8}	
			592.4	.01810	.905	5.0×10^{-8}	
			619.1	.01930	.965	3.6×10^{-8}	
			640.3	.01960	.980	4.9×10^{-8}	
			664.3	.02010	1.005	4.7×10^{-8}	
			688.2	.02040	1.02	4.0×10^{-8}	
Specimen S-9							

Test in progress

TABLE III

CREEP TEST DATE, STRESS-RELIEVED TZM FORGED DISC, TESTED AT
2000°F (1093°C), 10,000 PSI (6.89 x 10⁷N/m²)

Time (Minutes)	Length Change			Time (Hours)	Length Change		
	ΔL (in) (2" G. L.)	Creep (%)	Pressure (Torr)		ΔL (in) (2" G. L.)	Creep (%)	Pressure (Torr)
1	-.00030	-.015	1.6 x 10 ⁻⁷	352.3	.00105	.0525	1.9 x 10 ⁻⁹
2	-.00015	-.0075		376.6	.00105	.0525	1.3 x 10 ⁻⁹
3	-.00015	-.0075		400.8	.00110	.055	2.1 x 10 ⁻⁹
4	-.00015	-.0075		457.5	.00130	.065	1.3 x 10 ⁻⁹
5	-.00005	-.0025		472.6	.00135	.0675	1.2 x 10 ⁻⁹
6	.00010	.005		497.0	.00140	.070	1.2 x 10 ⁻⁹
7	.00005	.0025		520.2	.00145	.0725	1.2 x 10 ⁻⁹
8	.0000	.000		544.1	.00145	.0725	1.2 x 10 ⁻⁹
9	.00010	.005		568.2	.00150	.075	1.2 x 10 ⁻⁹
10	.00005	.0025		640.2	.00155	.0775	1.5 x 10 ⁻⁹
15	.00005	.0025		664.3	.00170	.085	9.2 x 10 ⁻¹⁰
20	.00010	.005		712.3	.00175	.0875	7.4 x 10 ⁻¹⁰
25	.00005	.0025		736.3	.00170	.085	7.6 x 10 ⁻¹⁰
30	.00000	.0000		810.2	.00175	.0875	1.3 x 10 ⁻⁹
45	.00010	.005		832.2	.00185	.0925	9.0 x 10 ⁻¹⁰
60	.00015	.0075		856.1	.0018	.090	9.2 x 10 ⁻¹⁰
75	.00005	.0025		880.1	.0020	.100	7.5 x 10 ⁻¹⁰
90	.00010	.005		904.0	.00205	.1025	7.7 x 10 ⁻¹⁰
<u>(Hours)</u>				982.8	.00210	.105	8.1 x 10 ⁻¹⁰
20.0	-.00005	-.0025	6.7 x 10 ⁻⁸	1000.2	.00210	.105	7.6 x 10 ⁻¹⁰
74.7	.00030	.015	1.0 x 10 ⁻⁸	1024.2	.00210	.105	8.2 x 10 ⁻¹⁰
125.7	.00100	.05	1.4 x 10 ⁻⁸	1048.3	.00220	.110	8.0 x 10 ⁻¹⁰
136.5	.00080	.04	1.0 x 10 ⁻⁸	1072.1	.00220	.110	1.3 x 10 ⁻⁹
160.3	.00085	.0425	5.7 x 10 ⁻⁹	1114.0	.00230	.115	6.9 x 10 ⁻¹⁰
184.4	.00085	.0425	4.4 x 10 ⁻⁹	1168.3	.00235	.1175	6.9 x 10 ⁻¹⁰
208.5	.00090	.045	4.0 x 10 ⁻⁹	1192.2	.00240	.120	6.6 x 10 ⁻¹⁰
232.2	.00090	.045	3.4 x 10 ⁻⁹	1216.4	.00250	.125	6.4 x 10 ⁻¹⁰
304.4	.00100	.05	2.2 x 10 ⁻⁹	1240.2	.00250	.125	5.8 x 10 ⁻¹⁰
328.3	.00100	.05	2.0 x 10 ⁻⁹	1336.2	.00250	.125	5.9 x 10 ⁻¹⁰

Specimen B-3

TABLE III (Cont.)

<u>Time (Hours)</u>	<u>Length Change L (in) (2" G. L.)</u>	<u>Creep (%)</u>	<u>Pressure (Torr)</u>
Specimen B-3	1360.3	.00250	.125
	1384.3	.00250	.125
	1408.5	.00250	.125
	1480.3	.00260	.13
	1552.2	.00270	.135
	1648.3	.00320	.16
	1696.4	.00320	.16
	1720.6	.00320	.16
	1816.4	.00320	.16
	1864.2	.00315	.157
	1912.3	.00325	.162
	1984.3	.00330	.165
	2032.2	.00335	.167
	2080.1	.00340	.17

Test in progress

TABLE IV

CREEP TEST DATA, TZM FORGED DISC, RECRYSTALLIZED AT 2840°F (1566°C), FOR1 HOUR, TESTED AT 2000°F (1093°C) 10,000 PSI (6.89 x 10⁷ N/m²)

Time (Minutes)	Length Change			Time (Hours)	Length Change		
	Δ L (in) 2" G. L.	Creep (%)	Pressure Torr		Δ L (in) 2" G. L.	Creep (%)	Pressure (Torr)
1	.00030	.015	6.0 x 10 ⁻¹⁰	16.6	.00040	.020	6.2 x 10 ⁻⁹
2	.00045	.0225		40.7	.00055	.0275	3.1 x 10 ⁻⁹
3	.00040	.020		114.5	.00095	.0475	5.1 x 10 ⁻⁹
4	.00030	.015		136.5	.00105	.0525	3.0 x 10 ⁻⁹
5	.00025	.0125		160.4	.00120	.060	3.4 x 10 ⁻⁹
6	.00015	.0075		184.4	.00130	.065	2.4 x 10 ⁻⁹
7	.00015	.0075		208.3	.00140	.070	2.3 x 10 ⁻⁹
8	.00010	.005		287.5	.00180	.090	2.2 x 10 ⁻⁹
9	.00020	.010		304.7	.00235	.1175	1.8 x 10 ⁻⁹
10	.00030	.015		328.7	.00305	.2525	1.6 x 10 ⁻⁹
11	.00030	.015		336.5	.00485	.2425	
12	.00025	.0125		352.6	.00515	.2575	1.6 x 10 ⁻⁹
13	.00030	.015		374.4	.00525	.2625	2.2 x 10 ⁻⁹
14	.00030	.015		448.3	.00525	.2625	1.4 x 10 ⁻⁹
15	.00035	.0175		472.7	.00525	.2625	1.0 x 10 ⁻⁹
20	.00045	.0225		496.6	.00530	.265	1.0 x 10 ⁻⁹
25	.00050	.025		520.8	.00535	.2675	1.6 x 10 ⁻⁹
30	.00055	.0275		544.6	.00565	.2825	1.2 x 10 ⁻⁹
35	.00050	.025		616.6	.00560	.280	6.2 x 10 ⁻¹⁰
40	.00050	.025		640.5	.00555	.2775	7.2 x 10 ⁻¹⁰
45	.00050	.025		664.6	.00560	.280	7.2 x 10 ⁻¹⁰
60	.00050	.025		688.7	.00560	.280	7.3 x 10 ⁻¹⁰
				712.8	.00565	.2825	1.3 x 10 ⁻⁹
				784.6	.00560	.280	5.6 x 10 ⁻¹⁰
				808.7	.00555	.277	5.5 x 10 ⁻¹⁰
				832.6	.00565	.282	5.8 x 10 ⁻¹⁰
				856.6	.00575	.287	5.8 x 10 ⁻¹⁰
				880.6	.00580	.290	6.6 x 10 ⁻¹⁰
				952.6	.00480	.240	4.3 x 10 ⁻¹⁰
				977.0	.00485	.242	1.4 x 10 ⁻⁹
				1000.7	.00485	.242	4.2 x 10 ⁻¹⁰
				1025.0	.00505	.252	7.2 x 10 ⁻¹⁰
				1049.3	.00440	.220	4.6 x 10 ⁻¹⁰
				1120.8	.00465	.232	3.2 x 10 ⁻¹⁰
				1145.2	.00485	.242	4.3 x 10 ⁻¹⁰
				1168.5	.00505	.252	5.0 x 10 ⁻¹⁰
				1192.7	.00515	.257	8.0 x 10 ⁻¹⁰
				1216.6	.00515	.257	3.0 x 10 ⁻¹⁰
				1288.6	.00510	.255	1.6 x 10 ⁻¹⁰
				1315.5	.00525	.262	2.2 x 10 ⁻¹⁰
				1336.5	.00525	.262	1.5 x 10 ⁻¹⁰
				1360.7	.00520	.260	1.6 x 10 ⁻¹⁰
				1384.5	.00525	.262	2.1 x 10 ⁻¹⁰

Test in progress

TABLE V

CREEP TEST DATA, TZC PLATE, RECRYSTALLIZED AT 3092°F (1700°C) FOR 1 HOUR
TESTED AT 2000°F, 14,000 PSI

Time (Minutes)	Length Change ΔL (in) (2" G. L.)	Creep (%)	Pressure (Torr)	Time (Hours)	Length Change ΔL (in) (2" G. L.)	Creep (%)	Pressure (Torr)
1	-.00005	-.0025	5.3×10^{-9}	2	.00005	.0025	5.3×10^{-8}
2	-.00020	-.01		19	.00010	.005	2.8×10^{-8}
3	-.00020	-.01		43.3	.00010	.005	2.0×10^{-8}
4	-.00020	-.01		117.1	.00005	.0025	1.4×10^{-8}
5	-.00015	-.0075		139.1	.00005	.0025	9.6×10^{-10}
6	-.00015	-.0075		162.9	.00010	.005	9.1×10^{-10}
7	-.00010	-.005		187.0	.00010	.005	7.7×10^{-10}
8	-.00010	-.005		210.8	.00015	.0075	7.3×10^{-9}
9	-.00005	.0025		307.3	.00005	.0025	6.0×10^{-9}
10	.00005	.0025		331.1	.00015	.0075	5.7×10^{-9}
11	.00010	.005		355.1	.00015	.0075	4.7×10^{-9}
12	.00005	.0025		378.9	.00005	.0025	3.2×10^{-9}
13	.00010	.005		450.8	.00005	.0025	4.4×10^{-9}
14	.00005	.0025		475.2	.00015	.0075	2.4×10^{-9}
15	.00005	.0025		523.3	.00010	.005	2.8×10^{-9}
20	.00005	.0025					
25	.00005	.0025					Test discontinued due to low creep rate.
30	.00010	.005					
35	.00010	.005					
40	.00005	.0025					
45	.00010	.005					
60	.00010	.005					

Specimen B-5

TABLE VI

CREEP TEST DATA, TZC PLATE, RECRYSTALLIZED AT 3092°F (1700°C) FOR 1 HOUR, TESTED
AT 2000°F (1093°C), 20,000 PSI (1.38×10^8 N/m²)

<u>Time (Hours)</u>	<u>Length Change ΔL (in) (2" G. L.)</u>	<u>Creep (%)</u>	<u>Pressure (Torr)</u>
62.0	.00015	.0075	2×10^{-9}
85.9	.00035	.0175	1.8×10^{-9}
110.0	.00040	.020	2.2×10^{-9}
134.0	.00045	.022	1.6×10^{-9}
158.1	.00050	.025	2.2×10^{-9}
230.0	.00060	.030	2.2×10^{-9}
254.0	.00050	.025	2.2×10^{-9}

Test discontinued to reload at higher stress

TABLE VII

CREEP TEST DATA, TZC PLATE, RECRYSTALLIZED AT 3092°F (1700°C) FOR 1 HOUR, TESTED
AT 2000°F (1093°C), 22,000 PSI (1.52×10^8 N/m²)

<u>Time</u> <u>(Hours)</u>	<u>Length Change</u> <u>ΔL (in)</u>	<u>Creep</u> <u>(%)</u>	<u>Pressure</u> <u>(Torr)</u>
19.0	.00005	.0025	2.1×10^{-9}
43.0	.00000	.010	2.2×10^{-9}
67.1	.00035	.017	2.2×10^{-9}
139.0	.00045	.0225	1.6×10^{-9}
163.4	.00045	.0225	1.7×10^{-9}
187.1	.00045	.0225	1.9×10^{-9}
211.4	.00040	.020	2.2×10^{-9}
307.2	.00045	.0225	2.0×10^{-9}
331.5	.00040	.020	1.8×10^{-9}
354.9	.00040	.020	1.8×10^{-9}
379.1	.00025	.0125	1.6×10^{-9}
475.1	.00045	.0225	1.4×10^{-9}
547.1	.00035	.0175	1.3×10^{-9}
570.9	.00050	.025	1.4×10^{-9}

Test discontinued to use unit for another test

TABLE VIII

CREEP TEST DATA, TZC PLATE, RECRYSTALLIZED AT 3092°F (1700°C) FOR 1 HOUR, TESTED
AT 2000°F (1093°C), 20,000 PSI (1.38×10^8 N/m²)

Time (Minutes)	Length Change			Time (Hours)	Length Change		
	ΔL (in) (2" G.T.)	Creep (%)	Pressure (Torr)		ΔL (in) (2" G.T.)	Creep (%)	Pressure (Torr)
1	- 0	0	1.2×10^{-9}	17.2	.00040	.020	2.6×10^{-9}
2	-.00020	-.010		41.3	.00060	.030	2.2×10^{-9}
3	-.00040	-.020		65.2	.00080	.040	2.2×10^{-9}
4	-.00030	-.015		89.2	.00110	.055	2.0×10^{-9}
5	-.00030	-.015		161.3	.00105	.0525	1.3×10^{-9}
10	-.00005	-.0025		185.7	.00125	.0625	1.7×10^{-9}
20	+.00005	+.0025		209.1	.00120	.060	1.7×10^{-9}
25	+.00010	+.005		233.2	.00120	.060	1.7×10^{-9}
30	+.00010	+.005		257.3	.00125	.0625	1.3×10^{-9}
60	+.00005	+.0025		329.2	.00135	.065	2.0×10^{-9}
				355.9	.0014	.070	3.2×10^{-9}
Specimen B-9				377.0	.0014	.070	3.4×10^{-9}
				401.1	.0014	.070	2.4×10^{-9}
				425.0	.00145	.0725	1.5×10^{-9}

Test in progress

TABLE IX

CREEP TEST DATA, TZC PLATE, RECRYSTALLIZED AT 3092°F (1700°C) FOR 1 HOUR, TESTED
AT 2200°F (1204°C), 18,000 PSI (1.24×10^8 N/m²)

Time (Minutes)	Length Change			Time (Hours)	Length Change		
	L ₁ (in) (2" G.T.)	Creep (%)	Pressure (Torr)		L ₂ (in) (2" G. L.)	Creep (%)	Pressure (Torr)
1	-.00005	-.0025	3×10^{-9}	64.3	.00110	.055	1.1×10^{-9}
2	0	0		88.4	.00120	.060	3.2×10^{-9}
3	0	0		112.5	.00140	.070	2.7×10^{-9}
4	0	0		136.4	.00165	.082	3.4×10^{-9}
5	0	0		160.4	.00215	.107	2.0×10^{-9}
10	.00005	.0025		232.5	.00255	.127	2.7×10^{-9}
20	.00010	.005		256.9	.00270	.135	1.8×10^{-9}
30	.00005	.0025		280.4	.00280	.140	1.8×10^{-9}
40	.00010	.005		304.3	.00310	.155	9.6×10^{-10}
60	.00005	.0025		328.5	.00325	.162	2.0×10^{-9}
				400.4	.00365	.182	1.7×10^{-9}
				427.1	.00390	.195	5.5×10^{-10}
				448.2	.00510	.255	1.8×10^{-9}
				472.3	.00580	.290	1.6×10^{-9}
				496.3	.00610	.305	1.4×10^{-9}
Specimen B-8							

Test in progress

TABLE X

CREEP TEST DATA, STRESS-RELIEVED AS-30 PLATE, TESTED AT 2000°F (1093°C)
11,000 PSI (7.58 x 10⁷ N /m²)

Time (Minutes)	Length Change		Pressure (Torr)	Time (Hours)	Length Change		Creep (%)	Pressure (Torr)
	ΔL (in) (?'' G.T.)	Creep (%)			ΔL (in) (?'' G.L.)			
1	-	-		?	.00020		.010	
2	.00020	.010	1.6 x 10 ⁻⁷	2.5	.00040		.020	1.1 x 10 ⁻⁷
3	.00020	.010		15.9	.00165		.0825	1.0 x 10 ⁻⁸
4	.00000	0		39.8	.0017		.085	7.8 x 10 ⁻⁹
5	.00005	.0025		63.7	.00210		.105	5.6 x 10 ⁻⁹
6	.00000	0		87.4	.00235		.1175	5.5 x 10 ⁻⁹
7	.00005	.0025		159.4	.00315		.1575	3.8 x 10 ⁻⁹
8	.00005	.0025		183.7	.00400		.200	
9	.00005	.0025		207.6	.00400		.205	2.4 x 10 ⁻⁹
10	.00005	.0025		231.9	.00410		.215	2.6 x 10 ⁻⁹
11	.00005	.0025		255.7	.00430		.280	2.1 x 10 ⁻⁹
12	-	-		327.7	.00560		.287	1.6 x 10 ⁻⁹
13	.00005	.0025		351.6	.00575		.302	1.6 x 10 ⁻⁹
14	-	-		375.6	.00605		.327	1.2 x 10 ⁻⁹
15	.00005	.0025		399.7	.00655		.342	1.7 x 10 ⁻⁹
20	.00005	.0025		423.8	.00685		.397	5.2 x 10 ⁻⁹
30	.00005	.0025		495.7	.00795		.405	5.8 x 10 ⁻⁹
45	.00010	.005		519.6	.00810		.492	6.4 x 10 ⁻⁹
60	.00020	.010		543.6	.00985		.510	7.4 x 10 ⁻⁹
				567.7	.01020		.535	6.5 x 10 ⁻⁹
				591.7	.01070		.597	2.0 x 10 ⁻⁹
				663.6	.01195		.617	3.4 x 10 ⁻⁹
				688.0	.01235		.630	2.5 x 10 ⁻⁹
				711.8	.01260		.642	3.3 x 10 ⁻⁹
				736.1	.01285		.670	8.6 x 10 ⁻⁹
				760.3	.01340		.745	1.0 x 10 ⁻⁸
				856.2	.01490		.755	1.0 x 10 ⁻⁸
				879.6	.01510		.765	3.7 x 10 ⁻⁹
				903.7	.01530		.800	8.5 x 10 ⁻⁹
				927.7	.01600		.842	7.2 x 10 ⁻⁹
				999.7	.01685		.857	2.4 x 10 ⁻⁹
				1026.5	.01715		.890	7.9 x 10 ⁻⁹
				1047.6	.01780		.897	6.0 x 10 ⁻⁹
				1071.8	.01795		.925	2.8 x 10 ⁻⁹
				1095.5	.01850		1.01	8.6 x 10 ⁻⁹
				1167.7	.02020			5.5 x 10 ⁻⁹
				1191.9	.02035			

Specimen B-6 Test terminated after reaching 1% creep

TABLE XI

CREEP TEST DATA, STRESS-RELIEVED AS-30 PLATE, TESTED AT 2200°F (1204°C)
 8000 PSI (5.51×10^7 N/m²)

Time (Minutes)	Length Change			Time (Hours)	Length Change		
	ΔL (in) (2" G. L.)	Creep (%)	Pressure (Torr)		ΔL (in) (2" G. L.)	Creep (%)	Pressure (Torr)
1	.00025	.0125	6.7×10^{-9}	2.2	.00060	.030	6.7×10^{-9}
2	-	-		13.7	.00195	.0975	1.4×10^{-7}
3	-	-		46.0	.00455	.2275	2.7×10^{-8}
4	.00015	.0075		61.8	.00640	.320	1.5×10^{-8}
5	.00025	.0125		85.6	.00785	.3925	1.2×10^{-8}
6	-	-		157.7	.01405	.7025	4.1×10^{-9}
7	.00025	.0125		181.6	.01620	.810	4.2×10^{-9}
8	.00040	.020		205.6	.01840	.920	3.3×10^{-9}
9	.00035	.0175		229.7	.02050	1.025	3.2×10^{-9}
10	.00035	.0175					
11	.00040	.020					
12	.00045	.0225					
13	.00045	.0225					
14	.00045	.0225					
15	.00045	.0225					
20	.00045	.0225					
30	.00040	.020					
45	.00040	.020					
60	.00045	.0225					

Test discontinued after reaching 1% creep.

Specimen B-7

EXTERNAL DISTRIBUTION

National Aeronautics and Space Administration
Washington, D. C. 20546
Attn: Walter C. Scott
Attn: James J. Lynch (RN)
Attn: George C. Deutsch (RR)

National Aeronautics and Space Administration
Scientific and Technical Information Facility
Box 5700
Bethesda, Maryland 21811

2 copies + 2 reproducible

National Aeronautics and Space Administration
Ames Research Center
Moffet Field, California 94035
Attn: Librarian

National Aeronautics and Space Administration
Goddard Space Flight Center
Greenbelt, Maryland 20771
Attn: Librarian

National Aeronautics and Space Administration
Langley Research Center
Hampton, Virginia 23365
Attn: Librarian

National Aeronautics and Space Administration
Manned Spacecraft Center
Houston, Texas 77001
Attn: Librarian

National Aeronautics and Space Administration
George C. Marshall Space Flight Center
Huntsville, Alabama 35812
Attn: Librarian

National Aeronautics and Space Administration
Jet Propulsion Laboratory
4800 Oak Grove Drive
Pasadena, California 91103
Attn: Librarian

National Aeronautics and Space Administration
21000 Brookpark Road
Cleveland, Ohio 44135

Attn: Librarian

Dr. Bernard Lubarsky (SPSD) MS 86-1
Roger Mather (NPTB)
G. M. Ault MS 105-1
Joe Joyce (NPTB) MS 86-5
Paul Moorhead (NPTB)
John Creagh MS 86-5
John E. Dilley (SPSPS) MS 85-1
Norman T. Musial MS 77-1
Thomas Strom MS 5-8
T. A. Moss (NPTB) MS 86-5
Dr. Louis Rosenblum (MSD) (106-1)
R. Titran MS 105-1
R. Hall MS 105-1
George Tulisiak MS 14-1

10 copies

National Aeronautics and Space Administration
Western Operations Office
150 Pico Boulevard
Santa Monica, California 90406
Attn: Mr. John Keeler

National Bureau of Standards
Washington 25, D. C.
Attn: Librarian

Aeronautical Systems Division
Wright-Patterson Air Force Base, Ohio
Attn: Charles Armbruster (ASRPP-10)
T. Cooper
Librarian
John L. Morris
H. J. Middendorp ASN RG 33143

Army Ordnance Frankford Arsenal
Bridesburg Station
Philadelphia 37, Pennsylvania
Attn: Librarian

I. I. T. Research Institute
10 W. 35th Street
Chicago, Illinois 60616

Atomics International
8900 DeSoto Avenue
Canoga Park, California

AVCO
Research and Advanced Development Department
201 Lowell Street
Wilmington, Massachusetts
Attn: Librarian

Babcock and Wilcox Company
Research Center
Alliance, Ohio
Attn: Librarian

Battelle Memorial Institute
505 King Avenue
Columbus, Ohio
Attn: C. M. Allen
Attn: Librarian

The Bendix Corporation
Research Laboratories Division
Southfield, Detroit 1, Michigan
Attn: Librarian

The Boeing Company
Seattle, Washington
Attn: Librarian

Brush Beryllium Company
Cleveland, Ohio
Attn: Librarian

Carborundum Company
Niagara Falls, New York
Attn: Librarian

Bureau of Ships
Department of the Navy
Washington 25, D. C.
Attn: Librarian

Bureau of Weapons
Research and Engineering
Material Division
Washington 25, D. C.
Attn: Librarian

U. S. Naval Research Laboratory
Washington 25, D. C.
Attn: Librarian

Advanced Technology Laboratories
Division of American Standard
369 Whisman Road
Mountain View, California
Attn: Librarian

Aerojet General Nucleonics
P. O. Box 77
San Ramon, California
Attn: Librarian

AiResearch Manufacturing Company
Sky Harbor Airport
402 South 36th Street
Phoenix, Arizona
Attn: Librarian
E. A. Kovacevich

AiResearch Manufacturing Company
9851-9951 Sepulveda Boulevard
Los Angeles 45, California
Attn: Librarian

Chance Vought Aircraft Inc.
P. O. Box 5907
Dallas 22, Texas
Attn: Librarian

Clevite Corporation
Mechanical Research Division
540 East 105th Street
Cleveland 8, Ohio
Attn: Mr. N. C. Beerli,
Project Administrator

Climax Molybdenum Company of Michigan
Detroit, Michigan
Attn: Librarian

Convair Astronautics
5001 Kerrny Villa Road
San Diego 11, California
Attn: Librarian

Curtiss-Wright Corporation
Research Division
Tuehanna, Pennsylvania
Attn: Librarian

E. I. dePont de Nemours and Company, Inc.
Wilmington 98, Delaware
Attn: W. E. Lusby, Jr.
Defense Contracts Supervisor

Electro-Optical Systems, Incorporated
Advanced Power Systems Division
Pasadena, California
Attn: Librarian

Fansteel Metallurgical Corporation
North Chicago, Illinois
Attn: Librarian
Attn: Henry L. Kohn

Ford Motor Company
Aeronutronics
Newport Beach, California
Attn: Librarian

General Atomic
John Jay Hopkins Laboratory
P. O. Box 608
San Diego 12, California
Attn: Librarian

General Electric Company
Atomic Power Equipment Division
P. O. Box 1131
San Jose, California

General Electric Company
Missile and Space Vehicle Dept.
3198 Chestnut Street
Philadelphia 4, Pennsylvania

General Electric Company
Vallecitos Atomic Laboratory
Pleasanton, California
Attn: Librarian

General Electric Company
Evendale, Ohio 45215
FPD Technical Information Center
Bldg. 100, Mail Drop F-22

General Electric Company
Reentry Systems Dept.
Cincinnati, Ohio 45215
Attn: Dr. J. W. Semmel
E. E. Hoffman

2 copies

General Dynamics/Fort Worth
P. O. Box 748
Fort Worth, Texas
Attn: Librarian

General Motors Corporation
Allison Division
Indianapolis 6, Indiana
Attn: Librarian

Hamilton Standard
Division of United Aircraft Corporation
Windsor Locks, Connecticut
Attn: Librarian

Hughes Aircraft Company
Engineering Division
Culver City, California
Attn: Librarian

Lockheed Missiles and Space Division
Lockheed Aircraft Corporation
Sunnyvale, California
Attn: Librarian

Marquardt Aircraft Company
P. O. Box 2013
Van Nuys, California
Attn: Librarian

The Martin Company
Baltimore 3, Maryland
Attn: Librarian

The Martin Company
Nuclear Division
P. O. Box 5042
Baltimore 20, Maryland
Attn: Librarian

Martin Marietta Corporation
Metals Technology Laboratory
Wheeling, Illinois

Massachusetts Institute of Technology
Cambridge 39, Massachusetts
Attn: Librarian

Materials Research and Development
Manlabs, Inc.
21 Erie Street
Cambridge 39, Massachusetts

Materials Research Corporation
Orangeburg, New York
Attn: Librarian

McDonnell Aircraft
St. Louis, Missouri
Attn: Librarian

MSA Research Corporation
Callery, Pennsylvania
Attn: Librarian

National Research Corporation
70 Memorial Drive
Cambridge 42, Massachusetts
Attn: Librarian

North American Aviation
Los Angeles Division
Los Angeles 9, California
Attn: Librarian

Norton Company
Worcester, Massachusetts
Attn: Librarian

Pratt & Whitney Aircraft
400 Main Street
East Hartford 8, Connecticut
Attn: Librarian

Pratt & Whitney Aircraft
CANEL
P. O. Box 611
Middletown, Connecticut
Attn: Librarian

Republic Aviation Corporation
Farmingdale, Long Island, New York
Attn: Librarian

Solar
2200 Pacific Highway
San Diego 12, California
Attn: Librarian

Southwest Research Institute
8500 Culebra Road
San Antonio 6, Texas
Attn: Librarian

Rocketdyne
Canoga Park, California
Attn: Librarian

Superior Tube Company
Norristown, Pennsylvania
Attn: Mr. A. Bound

Sylvania Electric Products, Inc.
Chemical and Metallurgical
Towanda, Pennsylvania
Attn: Librarian

Temescal Metallurgical
Berkeley, California
Attn: Librarian

Union Carbide Corporation
Parma Research Center
Technical Information Service
P. O. Box 6116
Cleveland, Ohio 44101

Union Carbide Metals
Niagara Falls, New York
Attn: Librarian

Union Carbide Stellite Corporation
Kokomo, Indiana
Attn: Librarian

Union Carbide Nuclear Company
P. O. Box X
Oak Ridge, Tennessee
Attn: X-10 Laboratory Records Department 2 copies

United Nuclear Corporation
5 New Street
White Plains, New York
Attn: Librarian
Attn: Mr. Albert Weinstein,
Sr. Engineer

Universal Cyclops Steel Corporation
Refractomet Division
Bridgeville, Pennsylvania
Attn: C. P. Mueller

University of Michigan
Department of Chemical and Metallurgical Eng.
Ann Arbor, Michigan
Attn: Librarian

U. S. Atomic Energy Commission
Technical Reports Library
Washington 25, D. C.
Attn: J. M. O'Leary

U. S. Atomic Energy Commission
P. O. Box 1102
East Hartford, Connecticut
Attn: A. J. Alexander
CANEL Project Office

U. S. Atomic Energy Commission
Germantown, Maryland
Attn: Col. E. L. Douthett
SNAP 50/SPUR Project Office
Attn: H. Rochen
SNAP 50/SPUR Project Office
Attn: Socrates Christofer
Attn: Major Gordon Dicker
SNAP 50/SPUR Project Office

U. S. Atomic Energy Commission
Technical Information Service Extension
P. O. Box 62
Oak Ridge, Tennessee

3 copies

U. S. Atomic Energy Commission
Washington 25, D. C.
Attn: M. J. Whitman

Argonne National Laboratory
9700 South Cross Avenue
Argonne, Illinois
Attn: Librarian

Brookhaven National Laboratory
Upton Long Island, New York
Attn: Librarian

Oak Ridge National Laboratory
Oak Ridge, Tennessee
Attn: W. C. Thurber
Attn: Dr. A. J. Miller
Attn: Librarian

Office of Naval Research
Power Division
Washington 25, D. C.
Attn: Librarian

Vought Astronautics
P. O. Box 5907
Dallas 22, Texas
Attn: Librarian

Wah Chang Corporation
Albany, Oregon
Attn: Librarian

Westinghouse Electric Corporation
Astronuclear Laboratory
P. O. Box 10864
Pittsburgh, Pennsylvania 15236
Attn: R. Begley

Westinghouse Electric Corporation
Materials Manufacturing Division
RD No. 2 Box 25
Blairsville, Pennsylvania
Attn: Librarian
Attn: F. L. Orrell

Wolverine Tube Division
Calumet and Hecla, Inc.
17200 Southfield Road
Allen Park, Michigan
Attn: Mr. Eugene F. Hill

Wyman-Gordon Company
North Grafton, Massachusetts
Attn: Librarian

Westinghouse Electric
Aerospace Electrical Division
Lima, Ohio
Attn: Paul Kueser

Air Force Materials Laboratory
Research and Technology Division
Wright-Patterson Air Force Base, Ohio 45433
Attn: C. L. Harmsworth

Johns Hopkins University
Applied Physics Laboratory
8621 Sergia Avenue
Silver Spring, Maryland
Attn: Librarian

Pratt & Whitney Aircraft
Materials Development Laboratory
P.O. Box 611
Middletown, Connecticut
Attn: Librarian

RCA
Defense Electronics Products
Astro-Electronics Division
Princeton, New Jersey
Attn: Librarian

Lawrence Radiation Lab
Technical Information Division
Livermore, California 94550
Attn: W. Blake Myers

General Electric Co.
Nuclear Materials and Propulsion Operation
P.O. Box 15132
Cincinnati, Ohio 45215
Attn: 710 Reactor Systems Program Operation